

# Service Bulletin

DoC No: ADxC-28-SB-002
Issue: A
Date: 26-Mar-2015

### D4 Fascination (VLA)

## 1.) Applicability D4 Fascination, EASA.A.018, SN1002,1003,1004 This service bulletin must be implemented in all affected products/aircraft. If not, the product/aircraft may lose its airworthiness status and eventual warranties become invalid. 2.) Planning information 2.1) Parts/Products affected: WD4-43-029.A Aileron actuator lever/ spade support arm 2.2) Concurrent SB: N/A Following lethal accident a recent national AD on D4 and D4BK 2.3) Reason: microlight predecessor models was issued concerning the weld of the spade support arm. Although different in design and complying with CS-VLA the corresponding part on the D4 Fascination (VLA) is identified as potential cause of catastrophic 2.4) Subject: Inspection of the weld shall be performed using dye penetrant or magnetic particle method. 2.5) Effective date: March 26th 2015 First inspection within next 50 flight hours or next annual 2.6) Compliance time: inspection whichever comes first. Repetitive inspection every 3 years 2.7) Statement: The technical content of this document is approved under the authority of DOA ref. EASA. 21.411. 2.8) Manpower: 1 h 2.9) Mass data: Not affected 2.10) Electrical load data: 2.11) Software accomplishment summary: 2.12) References: Include this SB in the airplane maintenance manual accordingly. 2.13) Other publications effected: 3.) Material information

The technical content of this document is approved under the authority of DOA ref. EASA. 21J.411.			
Prepared: Basien	CVE: Kölmel	Approved: Kölmel	

Top coat paint (insignificant amount)

In case cracks are found contact AD&C

N/A

see 4.)

Dye penetrant

3.1) Material-cost – availability:

3.3) Material requirements per

aircraft: 3.4) Rework parts:

3.5) Special tooling:

3.2) Company support information:



Service Bulletin DoC No: ADxC-28-SB-002
Issue: A
Date: 26-Mar-2015

#### D4 Fascination (VLA)

# 4.) Accomplishment/Instructions

Instructions are valid for left AND right spade support

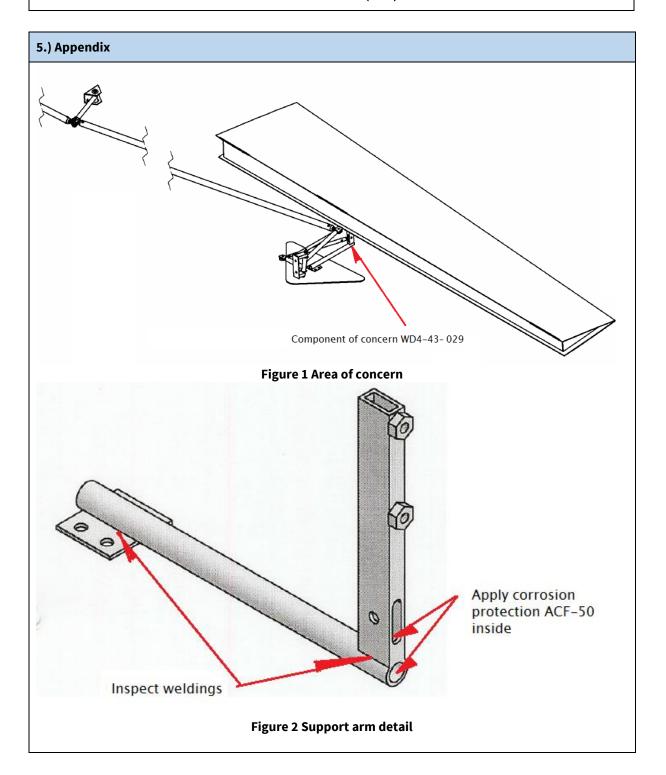
- A.) Inspect spade support/actuator arm attachment (no free play, no notable flexibility in the attachment area)
- B.) Visual inspect the weld areas for signs of cracks and/or corrosion, ref. Figure 1 & Figure 2.
- C.) Perform dye penetrant inspection on the welding shown in Figure 2. In case of need/doubt remove surface coating (local sanding or sandblasting). Note: removal of the support arm from the aileron is not foreseen as regular maintenance action. In case this is needed contact AD&C for further instructions.
- D.) If cracks are found the airplane cannot be released to service. In this case contact AD&C for further instructions.
- E.) If no cracks are found and no corrosion is present:
  - a. Recoat the support arm as appropriate per chapter 20-10-03 of the maintenance manual.
  - b. Apply corrosion protection "ACF-50" as indicated in Figure 2
  - c. Make appropriate entry in Airplane logbook and life cycle document of accomplishment of this SB.
  - d. Add repetition schedule of this inspection (every 3 years after first) to life cycle document of aircraft

The technical content of this document is approved under the authority of DOA ref. EASA. 21J.411.			
Prepared: Basien	CVE: Kölmel	Approved: Kölmel	



Service Bulletin DoC No: ADxC-28-SB-002
Issue: A
Date: 26-Mar-2015

# D4 Fascination (VLA)



The technical content of this document is approved under the authority of DOA ref. EASA. 21J.411.			
Prepared: Basien	CVE: Kölmel	Approved: Kölmel	