



## SUPPLEMENTAL TYPE CERTIFICATE

**10035295**

This Supplemental Type Certificate is issued by EASA, acting in accordance with Regulation (EC) No. 216/2008 on behalf of the European Community, its Member States and of the European third countries that participate in the activities of EASA under Article 66 of that Regulation and in accordance with Commission Regulation (EC) No. 1702/2003 to

### AIRCRAFT DESIGN & CERTIFICATION LTD

**REICHENSTEINSTRASSE 48  
69151 NECKARGEMUEND  
GERMANY**

and certifies that the change in the type design for the product listed below with the limitations and conditions specified meets the applicable Type Certification Basis and environmental protection requirements when operated within the conditions and limitations specified below:

**Original Product TC Number : EASA.A.024**

**TC Holder : AIRCRAFT INDUSTRIES A.S.**

**Model : L-13 "BLANÍK"**

#### **Description of Design Change:**

Modification and Inspection program for compliance with EASA AD-2010-0185-E

#### **EASA Certification Basis:**

Design:

Non-significant change

The Certification Basis of the original product remains applicable to this change

#### **Environmental Protection:**

The certificated noise and/ or emissions levels of the original product are unchanged and remain applicable to this certificate/ approval

See Continuation Sheet(s)

**For the European Aviation Safety Agency,**

**Date of issue: 14.06.2011**

  
European Aviation Safety Agency  
Roger HARDY  
Certification Manager  
General Aviation  
Tel: +49 221 89990 4006  
roger.hardy@easa.europa.eu

#### **Note:**

The following numbers are listed on the certificate:  
EASA current Project Number: 0010008723-001

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**Associated Technical Documentation:**

ADxC-39-001-AFM - 30 Mar 11 - Airplane Flight Manual supplement

ADxC-39-001-AMM - 11 June 11 - Sailplane Maintenance Manual supplement

ADxC-39-EO-013 Issue A 01. Apr 11 Template for installation of structural modification

ADxC-39-MDL 27-Apr-11 Master Document List

**Limitations:**

None

**Conditions:**

Prior to installation of this modification it must be determined that the interrelationship between this modification and any other previously installed modification and/ or repair will introduce no adverse effect upon the airworthiness of the product

- end -

**Note:**

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